Surge control of the electrically driven centrifugal compressor

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sum of the output power and the power consumed by the drive-compressor subsystem P_d^{el}

$$\eta_{\text{FC}}^{\text{E}} = \frac{P_{\text{FC}}^{\text{el}}}{P_{\text{FC}}^{\text{el}} + P_d^{\text{el}}} \quad (1)$$

From the FC static and dynamic models [1], [3], and [4], the requirements to the performance characteristics of the air supply equipment has been specified to provide a proper system operation. Required air supply profile has been derived using Faraday's law. In terrain conditions, the oxygen is normally taken from air. Then, considering the oxygen molar proportion in air, which equals 0.21, and the air molecular weight = 0.029 kg/mole and taking into account the cell number N_{cell} , the air mass flow is defined as

$$m = N_{\text{cell}} \frac{S_{\text{O}_2}}{0.21} \frac{I_{\text{FC}}}{4F} 0.029 \quad (2)$$

where Faraday constant $F = 96487$ C/mol, I_{FC} —FC current, (A), and S_{O_2} is the oxygen stoichiometry which denotes the ratio of the inlet and consumed amount of oxygen

$$S_{\text{O}_2} = \frac{m_{\text{O}_2}^{\text{in}}}{m_{\text{O}_2}^{\text{in}} - m_{\text{O}_2}^{\text{out}}} \quad (3)$$

where $m_{\text{O}_2}^{\text{in}}$ and $m_{\text{O}_2}^{\text{out}}$ —the inlet and outlet oxygen mass flows.

The FC air mass flow needs to be controlled rapidly and efficiently to avoid oxygen starvation and extend the lifetime of the stack [5]. Oxygen starvation occurs when the partial pressure of oxygen falls below a critical level in the FC cathode, especially at a high FC current. This phenomenon entails a rapid decrease in cell voltage, which in some cases can cause burn-through on the surface of a membrane.

This event is usually prevented either by the use of a large buffer, which allows to remove the current from the stack, or by providing the excess of oxygen stoichiometry, which normally results in the increased system complexity and extra auxiliary power consumption. Those problems could be avoided by keeping the oxygen stoichiometry at the constant level (in practice is approximately equal to two).

In this case, the operational area of the compression equipment is aimed to cover the required air mass flow levels (the correspondence of the FC and the compressor operation points is presented in Fig. 2). Besides, the operation of the compression equipment (and electrical drives) beyond the designed bounds results in the efficiency and stability lags (typical for centrifugal compressors) [6], [7].

Equation (1) for the system efficiency can be corrected with respect to an extension of the stable operational area of the compressor by introducing the term utilization κ_c of the compressor mass flow (and, respectively, its output power in case of constant pressure). It corresponds to the deviation from the constant stoichiometry. It denotes the ratio of the output power at the desired operation point to the instant one

$$\kappa_c = \frac{P_c^{\text{des}}}{P_c} \quad m^{\text{des}} \leq m \quad (\text{or } S_{\text{O}_2} \geq 2) \quad p = \text{const} \quad (4)$$

Fig. 2. FC stack and compressor operation points.

where the compressor output power P_c is

$$P_c = c_p T_{01} m \left[\left(\frac{p_2}{p_{01}} \right)^{\frac{\gamma-1}{\gamma}} - 1 \right] = P_d^{\text{el}} \eta_d \eta_c \quad (5)$$

where

- T_{01} ambient temperature, (K);
- p_{01} and p_2 ambient and outlet pressure, respectively, (Pa);
- m air mass flow, (kg/s);
- γ ratio of specific heats, (1.4 for air);
- η_d, η_c efficiencies of the electrical drive and the compressor, respectively.

Finally, the energy efficiency of the system with FC stack, air compressor, and electrical drive is expressed as

$$\eta_{\text{FC}}^{\text{E}} = \frac{\int_{t_1}^{t_2} V_{\text{FC}} I_{\text{FC}} dt}{\int_{t_1}^{t_2} V_{\text{FC}} I_{\text{FC}} dt + \int_{t_1}^{t_2} \frac{P_c^{\text{des}}}{\kappa_c \eta_d \eta_c} dt} \quad (6)$$

where $(t_1 - t_2)$ —time interval of an operation cycle.

The derived relation allows to express the system efficiency depending on the control performance and characteristics of the compressor. Graphical representation of the obtained equations is shown in Figs. 3 and 4. The maximum of the compressor efficiency is achieved in vicinity of the surge limit. Because the energy consumption is directly proportional to the mass flow, the system efficiency reaches its maximum at the point, where the utilization equals unity.

III. COMPRESSOR DYNAMIC MODEL AND SURGE CONTROL

The operation area of centrifugal compressors is limited by the so-called “surge” line, which defines the stability limit in the compressor map (lower graph in Fig. 2). Traditionally, the surge has been avoided using so-called “surge avoidance”

Fig. 3. Compressor efficiency ($\kappa_c - m^{\text{des}}$ coordinates).

schemes, which restrict the efficiency and operating range of the compressor (“control line” in Fig. 2).

In the presented approach of active surge control the electrical drive is used for surge control. This means that the compressor can function at a low mass flow and high pressure without surge providing the extension of the compressor operation area.

For an extensive description of the surge phenomenon and the control development, a compressor dynamic model is used, based on the Greitzer lumped parameters model [8], [9] (Fig. 5). The dynamic model is described by the system of differential and algebraic equations

$$\frac{dp_p}{dt} = \frac{a_0^2}{V_p}(m - m_t) \quad (7)$$

$$\frac{dm}{dt} = \frac{A_1}{L_c}(\Psi_c p_0 - p_p) \quad (8)$$

$$\frac{d\omega}{dt} = \frac{1}{J}(T_d - T_c) \quad (9)$$

$$T_c = m\sigma\omega r_2^2 \quad (10)$$

where

m_t	mass flow through the throttle, (kg/s);
a_0	inlet stagnation sonic velocity, (m/s);
A_1	area of impeller eye, (m^2);
L_c	duct length, (m);
J	spool moment of inertia, ($\text{kg} \cdot \text{m}^2$);
ω	compressor angular speed, (rad/s);
p_0	ambient pressure, (Pa);
p_p	plenum pressure, (Pa);
V_p	plenum volume, (m^3);
Ψ_c	compressor characteristics;
T_d	drive torque, ($\text{N} \cdot \text{m}$);
T_c	compressor torque, ($\text{N} \cdot \text{m}$);
$\sigma = 0.9$	slip factor;
r_2	external impeller radius, (m).

Fig. 4. System efficiency ($\kappa_c - m^{\text{des}}$ coordinates).

Fig. 5. Compressor, duct, plenum, and throttle system.

Fig. 6. Active surge control scheme.

The key feature of the active surge control is that the shaft angular speed becomes a function of the mass flow [10]

$$\Delta\omega = -c\Delta m \quad (11)$$

where $\Delta(\cdot)$ —deviation from an equilibrium, c —control gain.

The mass flow and the impeller angular speed are used in this feedback control law. In fact, the reliable sensors for transient mass flow are not available (or can be rather expensive). Thus, an observer for the estimation of the mass flow has been used based on the compressor model (7)–(10). Fig. 6 represents the general compression control system with the surge controller. The model parameters used in the design are presented in Table I.

TABLE I
COMPRESSOR MODEL PARAMETERS

$A_1, (m^2)$	$L_c, (m)$	$J, (kgm^2)$	$V_p, (m^3)$
0.0064	4.4	0.001	0.028

IV. INDUCTION MOTOR (IM) MODEL AND FIELD ORIENTED CONTROL

The high-speed IM has been selected as a prime mover for the FC compressor due to its well-known advantages like simple construction, low cost, high reliability, and high efficiency. For the IM control design, the equivalent two-phase motor model has been used which represents the three-phase high-speed IM. With the additional assumptions that the stator and rotor phase windings are electrically and magnetically symmetrical, the air gap is uniform and the field distribution is sinusoidal and additionally, the effects of magnetic saturation, eddy-currents, and hysteresis are neglected, the model equations for an IM can be expressed in the synchronously rotating dq frame [11]

$$\begin{aligned}
 v_{qs}^e &= R_s i_{qs}^e + \frac{d\lambda_{qs}^e}{dt} + \omega_e \lambda_{ds}^e \\
 v_{ds}^e &= R_s i_{ds}^e + \frac{d\lambda_{ds}^e}{dt} - \omega_e \lambda_{qs}^e \\
 0 &= R_r' i_{qr}^e + \frac{d\lambda_{qr}^e}{dt} + (\omega_e - \omega_r) \lambda_{dr}^e \\
 0 &= R_r' i_{dr}^e + \frac{d\lambda_{dr}^e}{dt} - (\omega_e - \omega_r) \lambda_{qr}^e \\
 \lambda_{qs}^e &= L_{ls} i_{qs}^e + L_m (i_{qs}^e + i_{qr}^e) \\
 \lambda_{ds}^e &= L_{ls} i_{ds}^e + L_m (i_{ds}^e + i_{dr}^e) \\
 \lambda_{qr}^e &= L_{lr}' i_{qr}^e + L_m (i_{qs}^e + i_{qr}^e) \\
 \lambda_{dr}^e &= L_{lr}' i_{dr}^e + L_m (i_{ds}^e + i_{dr}^e)
 \end{aligned} \quad (12)$$

where

- R_s, R_r' stator and rotor resistances, (Ω);
- L_{ls}, L_{lr}' stator and rotor leakage inductance, (H);
- L_m magnetizing or mutual inductances, (H);
- $\lambda_{ds}^e, \lambda_{qs}^e$ dq -components of stator flux linkage, (Wb);
- $\lambda_{dr}^e, \lambda_{qr}^e$ dq -components of rotor flux linkage, (Wb);
- v_{ds}^e, v_{qs}^e dq -components of stator voltage, (V);
- i_{ds}^e, i_{qs}^e dq -components of stator current, (A);
- i_{dr}^e, i_{qr}^e dq -components of rotor current, (A);
- $(\cdot)_r'$ rotor values referred to the stator;
- ω_e, ω_r electrical synchronous and rotor angular velocities, (rad/s).

The electromagnetic torque is expressed as

$$T_d = N_p \frac{3}{2} \frac{L_m}{L_r'} (i_{qs}^e \lambda_{dr}^e - i_{ds}^e \lambda_{qr}^e) \quad (14)$$

where N_p is the pole pair number.

The indirect rotor flux oriented control scheme has been implemented for the high-performance machine control. The requirement for the rotor flux orientation is that the stator current components i_{ds}^e and i_{qs}^e have to be in phase and quadrature with respect to the rotor flux. This can be realized by selecting a synchronously rotating dq -frame at speed ω_e whose d -axis is

Fig. 7. Laboratory experimental setup.

TABLE II
IM PARAMETERS

		Equivalent circuit parameters	
Output power, (kW)	30	L_m , (mH)	12
Terminal voltage line-to-line, (V)	500	L_{ls} , (mH)	0.27
Frequency, (Hz)	300	L_{lr}' , (mH)	0.27
Pole pair	1	R_s , (Ω)	0.07
Rated speed, (rpm)	17640	R_r' , (Ω)	0.16
Phase current, (A)	47	$Cos(\phi)$	0.82
Over-current ratio I_a/I_n	6.88	Motor efficiency, (%)	89

aligned with the rotor flux. This implies that the q -component of the rotor flux linkage λ_{qr}^e is equal to zero: $\lambda_{qr}^e = 0$. The machine is said to be rotor flux oriented.

With this method, the independent control of flux and torque is possible. By keeping the d -component of the rotor flux linkage λ_{dr}^e constant it is possible to achieve a fast torque response, resulting in fast speed or position responses.

From (14), the developed electromagnetic torque can now be written as

$$T_d = \frac{3}{2} \frac{L_m}{L_r'} N_p \lambda_{dr}^e i_{qs}^e \quad (15)$$

From the torque equation it can be seen that if the rotor flux linkage is not disturbed, adjusting the q -component of the stator current can independently control the electromagnetic torque.

V. ACTIVE SURGE CONTROL SIMULATION AND EXPERIMENTAL RESULTS

The described above active surge control of the compressor scheme has been implemented in the laboratory experimental set-up (Fig. 7) which has been installed in collaboration with the Energy Technology Group of the TU/e-Mechanical Engineering Department. The main components of the setup are: high-speed IM drive with the implemented field oriented control (FOC), power electronic converter (Semikron), centrifugal compressor (Vortech) with an embedded gearbox of ratio 1: 3.45, PVC piping system, and DAQ and control systems realized on the dSpace DS1104 controller board.

The parameters of the motor (close to the needs of the FC power system compressor) are specified in Table II. The IM equivalent circuit parameters have been obtained from the

Fig. 8. Simulation and experimental results—surge.

Fig. 9. Simulation and experimental results—surge control.

standard IM test [12] (dc, no-load and blocked-rotor) performed in the laboratory.

The combined simulation and experimentally obtained results of the system (7)–(9) are presented in Figs. 8 and 9. The first operating point (an intersection of the compressor and system characteristics) corresponds to the opened valve, and it is located within the stable area. When the valve is closed, the system characteristic is moved to the unstable area, where the deep surge is initialized. In case of active surge control application (11), the operation point is moved beyond the surge line but the process remains stable (Fig. 9). The effects of the surge control are shown in Fig. 10.

VI. FC SYSTEM SIMULATION RESULTS

After the experimental validation of the active surge suppression method by the drive speed control, three operating regimes of the whole system with the FC stack, compressor, and IM drive have been simulated and compared (Figs. 11–16). In the

Fig. 10. Experimental results—surge control.

Fig. 11. FCs—compressor simulation results in state coordinates (constant speed).

Fig. 12. FCs—compressor simulation results in state coordinates (variable speed).

Fig. 13. FCs—compressor simulation results in state coordinates (surge control).

Fig. 15. Drive input power.

Fig. 14. Mass flow.

first regime, the compression system operates at the constant speed and constant system characteristic (valve position). The second regime corresponds to the classical variable-speed operation also at the constant valve position. In the third regime, the speed and pressure are kept approximately constant, and the only mass flow is regulated according to the FC electrical load. The developed active surge control is applied in this regime. As it can be seen from the comparison of the results, the variable-speed operation provides higher efficiency at the low FC load, due to less energy consumption by the compressor. On the other hand, the inertia of the motor rotor and the compressor impeller slows down the speed, and, consequently, the mass flow response. In Fig. 12, a visible sudden drop in the FC voltage occurs, while the system operation point moves from the low-level load (o.p.1) to the maximum (o.p.2). In this operation mode an additional energy buffer has to be installed next to the FCs to reduce the FC current and compensate that delay. This solution could be beneficial when the energy buffer is already assumed for using in the system, e.g., in systems with FCs and supercapacitors [13]. However, the use of large

Fig. 16. System efficiency.

batteries is not always a favorite solution, especially in vehicles. The stabilization of the compressor allows to adjust the mass flow keeping very fast response. The time constant of the thermodynamic process excluding the impeller speed is defined by the geometrical parameters of the system (duct length and volume), and it is approximately one order lower, comparing to the rotor inertia.

VII. TIME-OPTIMAL CONTROL

The combined surge stabilization and the time-optimal control for the compression system has been developed where two inputs (the drive torque T_d and valve position k_v) are used for the control. For the simplification of the control design, the current work deals with the linear quadratic Gaussian (LQG) problem. The LQ stands for linear systems with quadratic performance criteria. The system may be affected by disturbances and measurement noise represented as stochastic processes, in particular, by Gaussian white noise. The optimal control design methods convert control system design problems to an

Fig. 17. LQG controller scheme.

optimization problem with time-domain performance criteria. With the LQG approach, the feedback controller (Fig. 17) coefficients are designed so that the following cost functional J_c is minimized under the “constraint” of the system dynamics $\dot{x}$

$$\min_{u(t)} J_c = \frac{1}{2} \int_0^\infty (y(t)^T Q y(t) + u(t)^T R u(t)) dt$$

subject to $\dot{x} = f(x, u) \quad x(t_0) = x_0$ (16)

where

$$x(t) = \begin{bmatrix} \dot{p}_p \\ \dot{m} \\ \dot{\omega} \end{bmatrix} \quad u(t) = \begin{bmatrix} k_t \\ 0 \\ T_d \end{bmatrix} \quad y(t) = \begin{bmatrix} p_p \\ m \\ \omega \end{bmatrix}$$

are state, input, and output vectors, respectively, and Q and R weighting matrices.

The deviation of x from the reference trajectory is penalized quadratically with a nonnegative (symmetric) weighting matrix Q in order to reflect different weights attached to different state components. With the choice of the weighting matrices Q and R a tradeoff between control performance (increasing Q) and low input energy (increasing R) can be achieved. The Q and R parameters have been tuned in order to obtain satisfactory behavior of the compression system. The final values of Q and R are: $Q = [10^{-12}, 1, 0.02]$, $R = [8.5 \cdot 10^9, 2 \cdot 10^3]$.

The linearized system around the equilibrium point is presented in the following way:

$$\Delta x(t) = x(t) - x(t)^0 \quad \Delta u(t) = u(t) - u(t)^0 \quad (17)$$

where $(\cdot)^0$ is equilibrium point and $\Delta(\cdot)$ is deviation from the equilibrium.

The state-space equation of the linearized system is, thus,

$$\begin{aligned} \Delta \dot{x}(t) &= A \Delta x(t) + B \Delta u(t) + G \Delta y(t) \nu(t) \\ \Delta z(t) &= C_z \Delta x(t) \\ \Delta y(t) &= C \Delta x(t) + \varpi(t) \end{aligned} \quad (18)$$

where

- $\Delta z(t)$ controllable output;
- A, B, C state-space matrices defined from the compressor model;
- $\nu(t)$ plant disturbances;

$\varpi(t)$ measurement noise;

C_z and G matrices defining controllable output and plant disturbances.

This control scheme allows to handle multiple-input–multiple-output (MIMO) system problems almost as easily as single-input–single-output (SISO) system problems.

An optimal trajectory is generated by choosing the input for $t \geq 0$ as

$$u(t) = -F_c \Delta x(t) \quad F_c = R^{-1} B^T X \quad (19)$$

where F_c is the state feedback gain matrix, X is the symmetric matrix, and the nonnegative-definite solution of the algebraic Riccati matrix equation is

$$A^T X + X A + C_z^T B C_z - X B R^{-1} B^T X = 0. \quad (20)$$

This solution requires that the state $x(t)$ should be fully accessible for measurement. But the state m of the system is not directly accessible and only the outputs ω and p are measured. Thus, the reconstruction of the state m is performed using an observer of the form

$$\dot{\hat{x}}(t) = A \hat{x}(t) + B \Delta u(t) + K (\Delta y(t) - C \hat{x}(t)) \quad (21)$$

where

$\Delta \hat{x}(t)$ estimate $x(t)$;

K observer gain matrix.

The state estimate error is defined as

$$e(t) = -\Delta \hat{x}(t) = A \Delta \hat{x}(t) + B \Delta u(t) + K (\Delta y(t) - C \Delta \hat{x}(t)). \quad (22)$$

The linearized closed-loop system is represented as

$$\begin{bmatrix} \Delta \dot{\hat{x}}(t) \\ \dot{e}(t) \end{bmatrix} = \begin{bmatrix} A - B F_c & -B F_c \\ 0 & A - K C \end{bmatrix} \begin{bmatrix} \Delta x(t) \\ e(t) \end{bmatrix} + \begin{bmatrix} G v(t) \\ G v(t) + K \varpi(t) \end{bmatrix}. \quad (23)$$

The eigenvalues of the open-loop system, the regulator, and the closed-loop system at the point, located to the left from the surge line, are presented in Fig. 18.

With this method, the different operation regimes can be realized, depending on the desired system requirements. The example of the improved variable-speed operation along the time-optimal trajectories is presented in Figs. 19 and 20.

VIII. CONCLUSION

This paper describes two innovative operation regimes of the electrically driven centrifugal compressor for FC applications. The main goal of these control methods is the improvement of the system efficiency during the reduced FC electrical load keeping the fast response on the FC demand. The proposed methods are based on the active surge suppression of the centrifugal compressor, which has been experimentally verified. Four operation regimes have been compared.

The constant-speed operating mode results in the fastest response, since there are no transients in the compression system.

Fig. 18. System eigenvalues.

Fig. 19. FC—compressor simulation results (LQG control).

Fig. 20. LQG control simulation results.

However, system efficiency is reduced due to the extra power consumption by the compressor.

In the classical variable-speed operation mode, the compressor also works entirely in the stable area, and it results in

the highest efficiency gain at the reduced load. Meanwhile, it causes the significant delay in the response due to the mechanical transient.

The tradeoff between efficiency and response time has been achieved while the active surge control is applied. The transient time is reduced by approximately 90% in comparison to the variable speed mode, but at cost of 4% efficiency. However, this method exhibits 5% higher efficiency, than the constant speed mode.

The time-optimal MIMO LQG controller has been proposed as the extension of the surge suppression scheme and variable speed operation. It provides the transition between the operating points along the time-optimal trajectories. As it was demonstrated, about 25% reduction of the transient time can be achieved comparing to the variable speed mode while keeping the same efficiency.

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